

In Orbit Timing Calibration of the Hard X-Ray Detector on Board Suzaku

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Abstract

The hard X-ray detector (HXD) on board the X-ray satellite Suzaku is designed to have a good timing capability with a 61 μ s time resolution. In addition to detailed descriptions of the HXD timing system, results of in-orbit timing calibration and performance of the HXD are summarized. The relative accuracy of time measurements of the HXD event was confirmed to have an accuracy of 1.9×10^{-9} s s⁻¹ per day, and the absolute timing was confirmed to be accurate to 360 μ s or better. The results were achieved mainly through observations of the Crab pulsar, including simultaneous ones with RXTE, INTEGRAL, and Swift.

Key words: space vehicles: instruments — X-rays:general – time

1. Introduction

The Hard X-ray Detector (HXD; Takahashi et al. 2007; Kokubun et al. 2007) on board Suzaku (Mitsuda et al. 2007) is a novel cosmic hard X-ray instrument working in the 10 – 600 keV range. In addition to the very low background level, the wide energy band, and the tightly collimated field of view, yet another important feature of the HXD is its timing capability. This is because time variability, periodic or aperiodic, is an important characteristic of compact cosmic X-ray sources, including neutron stars, white dwarfs, galactic black holes, and active galactic nuclei. Considering the wide range of time scales

involved in these variations, we have required the HXD to have a time resolution of 61 μ s, and a timing stability of the order of 10^{-9} typically for a day. To fulfill these goals, we carefully designed the timing system of the HXD, as summarized in section 2, and in more detail in Takahashi et al. (2007) and Kokubun et al. (2007).

The HXD sensor (HXD-S) consists of 16 identical “Well units” working as main detectors (Uchiyama et al. 2001), and 20 surrounding shield counters made of Bi₄Ge₃O₁₂ (BGO) scintillators (Yamaoka et al. 2006). Each Well unit detects incoming X-rays using 2 mm-thick silicon PIN diodes (Sugiho et al. 2001) and Gd₂SiO₅:Ce (hereafter GSO) scintillators, both surrounded by a common

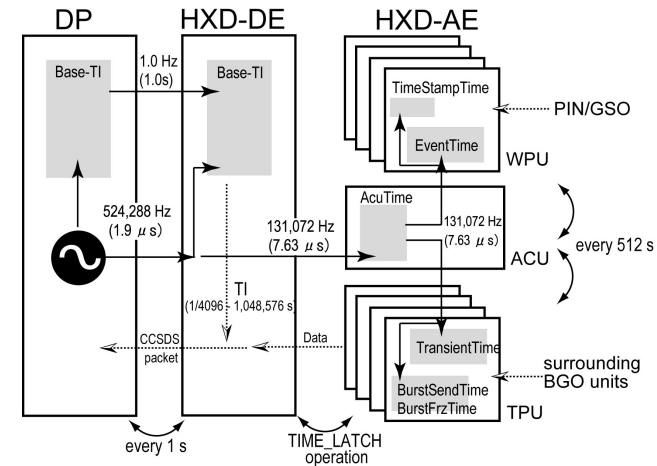


Fig. 1. Timing counters in the HXD electronics.

Before launch, we repeatedly tested and confirmed the HXD timing capability, first at each component level, then incorporating HXD-S, HXD-AE, and HXD-DE (Kokubun et al. 2003; Tashiro et al. 2002; Kawaharada et al. 2004; Terada et al. 2006; Ohno et al. 2006). The tests included confirming time assignments of single WEL-type events with specified time intervals (1, 10, 100, and 1000 s) from a time origin, detection of periodic signals (with periods of 1, 10, and 100 ms), and measuring time-interval distribution using random events. After the HXD was mounted on the spacecraft, another series of end-to-end tests were repeated incorporating spacecraft central clock, radio transmitters, and ground receivers.

Through the pre-launch verification, we confirmed that the HXD has correct timing function as designed. However, these tests were limited in many aspects. For example, the absolute timing measurement was performed only before the HXD was mounted on the spacecraft. The ground equipment used in these end-to-end tests was not identical to those actually used at the Suzaku tracking center, the Uchinoura Space Center (USC) in southern Japan. Furthermore, once the spacecraft is put into orbit, we must consider additional complications, such as signal delays due to air propagation to the ground station, and due to cable transmission from the antenna to signal receivers.

The present paper deals with in-orbit timing calibration of the HXD. After systematic errors affecting the time assignment are estimated in section 3, we analyze in section 4 highly stable periodic signals from four fast-rotating pulsars, including the Crab pulsar in particular. These tell us the relative accuracy and stability of the time measurements with the HXD. Finally, we verify the absolute timing accuracy of the HXD, by comparing the HXD measurements of the Crab pulsar against those with

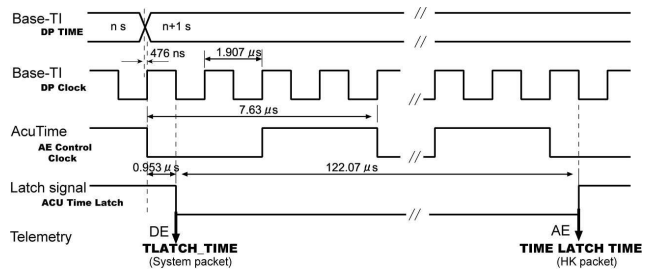


Fig. 2. A timing chart between HXD-AE and HXD-DE at the TIME LATCH operation. DP_TIME and DP_Clock are the signals from DP to HXD-DE to acknowledge time information of the spacecraft, and AE_Control_Clock is provided into HXD-AE from HXD-DE. The signal named ACU_Time_Latch is produced at a time-latch operation in HXD-DE, and is supplied into HXD-AE to verify timings between them.

other X-ray missions and with radio telescopes.

2. The Timing System of the HXD

2.1. Time Assignment of the Suzaku Data

All the Suzaku data acquired in orbit utilize a common timing system based on a 524,288 Hz quartz clock in the spacecraft central data processor (DP). Since the satellite has an orbit with an altitude of 568 km, an eccentricity of < 0.0002 , and an inclination of 31.4° , it has an orbital period of about 90 min, making 15 revolutions a day. Therefore, only 5 contacts from the ground station, USC, are available per day. All the data packets are stored into an on-board data recorder (DR) of 6 Gbits, and transferred via down link of 4.19 Mbps to ground during the satellite contacts. Therefore, the time assignment of the Suzaku data consists of two steps; the first is to calibrate time origin of the on-board clock against the Universal Time Coordinated (UTC) values during every ground contact, and the second is to determine the time when each telemetry packet was edited (usually outside the ground contacts), referring to the on-board clock counts after correction for their temperature-dependent drifts.

In order to afford the first step of the time assignment, the on-board clock counts, called “TI”, are attached to all the data packets (to be sent to telemetry) which are in so called CCSDS (Consultative Committee for Space Data System) format. Each TI has 32 bits, and covers 10^{20} s = 1,048,576 s \simeq 12 days with $1/4096$ s = 244μ s least significant bit. When the spacecraft makes ground contacts, this 32-bit values in TI format are calibrated with ground Rubidium clocks in UTC format with a stability of 10^{-11} s s $^{-1}$ month $^{-1}$. That is, all the relevant information, including the TIs and temperature of the clock, are gathered every 32 s into special packets called “timing CCSDS packets”, which are sent via S-Band real-time down link with the highest priority. A time delay inside the spacecraft from a packet edition in the DP to its acknowledgement by the spacecraft transmitter is fixed at 18.0μ s. Since the traveling time from the spacecraft to the ground station

Table 1. Summary of Timing Counters of the Hard X-ray Detector.

Name	Component	Length (bit)	Base (s)	Coverage in Telem. (s)	Telemetry Type/Packet (HK/Event)
Base-TI	DP / Spacecraft	39	$1/2^{19}(1.9\mu)$	$1/2^{12}(244\mu) - 2^{20}(1,048,576)$	HK /time packet
	HXD-DE	39	$1/2^{19}(1.9\mu)$	$1/2^{12}(244\mu) - 2^{20}(1,048,576)$	HK / HXD-HK
AcuTime	HXD-AE ACU	24	$1/2^{17}(7.63\mu)$	$1/2^{15}(30.5\mu) - 2^9(512)$	HK / HXD-HK
TimeStampTime	HXD-AE WPU	24	$1/2^{15}(30.5\mu)$	$1/2^{15}(30.5\mu) - 2^9(512)$	HK / HXD-HK
EventTime [†]	HXD-AE WPU	19	$1/2^{14}(61\mu)$	$1/2^{14}(61\mu) - 2^5(32)$	Event / HXD_WEL
TransientTime [‡]	HXD-AE TPU	24	$1/2^{15}(30.5\mu)$	$1/2^{15}(30.5\mu) - 2^9(512)$	Event / HXD_WAM
BurstSendTime	HXD-AE TPU	23	$1/2^6(15.6m)$	$1/2^6(15.6m) - 2^{18}(262144)$	Event / HXD_WAM
BurstFrzTime	HXD-AE TPU	32	$1/2^{15}(30.5\mu)$	$1/2^5(30.5m) - 2^{18}(262144)$	Event / HXD_WAM

[†] Time value tagged to each event which detected by a main unit of the HXD sensor.

[‡] Time value used for the HXD-WAM (Wide-band All-sky Monitor (Yamaoka et al. 2006)).

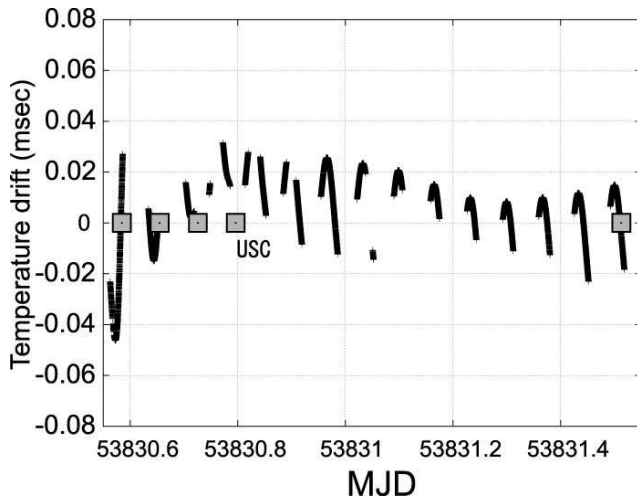


Fig. 3. An example of cumulative clock drifts, expected for the data taken on 2005 September. They are estimated by the measured DP temperature, and the temperature-dependent clock drift rate measured before launch. The horizontal axis shows the Modified Julian Day (MJD). Satellite contacts at the USC are also indicated.

depends on the distance between them, the spacecraft location is measured by ranging about twice a day, within an accuracy of 1 km along the line-of-sight, while about 10 km in three dimensions. Finally, a time delay within the ground station, from the receiver to the recorder, is also known to be less than 2 μ s. On each data packet received on ground, a “time stamps” is imprinted based on the Rubidium clocks with a 10 or 100 μ s timing resolution, depending on which of the 34m and 20m antenna systems at USC, respectively, is used to receive the data.

In the second step of the time assignment, TIs issued while the spacecraft is out of contacts are converted to UTC values through the TI vs. UTC relation, established in the first step using the closest ground contacts. Since the clock of Suzaku is not placed in a thermostatic environment, its temperature changes with the spacecraft attitude by about ± 20 K around ~ 290 K, leading to temperature-dependent clock drifts. However, this effect has been measured on ground and tabulated before the

launch, and the table is loaded to the on-board DP. The DP monitors the clock temperature, and calculates every 31.25 ms the corresponding integrated drift time, Y , in units of $1/4096/4096/100$ s = 0.6 ns. The values of Y are also recorded into the timing CCSDS packets. According to the pre-launch sea-level tests, TI values after the temperature correction have a stability of about 4×10^{-9} , which accumulates to about 25 μ s in one revolution.

2.2. Calculation of Arrival Time of HXD Events

Using the method described in section 2.1, we can determine the time values when individual telemetry packets are edited, but which we need to know is the arrival time of each event to the HXD sensor. Considering a mission requirement of a few hundred μ s resolution, we designed the HXD so that each event is tagged with its arrival time information with a resolution of $1/2^{14} = 61.0$ μ s nominally, and $1/2^{15} = 30.5$ μ s on condition (selectable by commands). However, because of severe limitation of the telemetry transfer rate, only 16 bytes information is available per event, which must include not only timing, but also 6 pulse-heights (4 PINs, GSO, and BGO), quality flags, trigger information, and hit-pattern information (Takahashi et al. 1998; Terada et al. 2006). Consequently, only 19 bits per event (hereafter “EventTime”) are available for the timing, which covers $1/2^{14} = 61.0$ μ to 32 seconds. Thus, we need to have a mechanism to relate “EventTime” to the spacecraft time in TI.

Figure 1 shows various timing counters in the HXD electronics, which are also summarized in table 1. The spacecraft clock (section 2.1) stays in the data processing unit (DP), and generates a timing clock of $1/2^{19} = 1.9$ μ s period. As shown there, the clock is first supplied to HXD-DE, then divided into the control module (Analog Control Unit, ACU) of HXD-AE, and finally supplied into eight analog modules (four Well-type detector Processing Unit, WPU, and four Transient detector Processing Unit, TPU) of HXD-AE.

HXD-DE always holds the latest TI value, supplied from DP every 2^{19} DP clocks (corresponding to 1 s). In HXD-AE, the time reference is provided by “AcuTime” in ACU, which is refreshed every 512 s by carry reset signals. The values of counters of “AcuTime”

and TI are synchronized, and cross-checked at the beginning of each observation via TIME_LATCH operation, where they are latched according to the timing chart in figure 2. The latched values of “AcuTime” and TI are sent to the telemetry, and stored as ‘HXD_AE_TM_LATCH_TM’ and ‘HXD_TLATCH_TIME’ columns in the HXD HK FITS file, respectively. This operation occurs typically once every one to two days. Besides, the eight analog precessing modules in HXD-AE (four WPUs and four TPUs) have such timing counters as “TimeStampTime”, “EventTime”, “TransientTime”, “BurstSendTime”, and “BurstFrzTime”, which are synchronized to the “AcuTime” signals.

As explained so far, an X-ray event detected with the HXD is given a UTC time through the following two steps; first to relate a value of 19-bit “EventTime” into the corresponding TI value, and then to convert it into the UTC value as described in section 2.1. In order to conduct the first step without any ambiguity due to scalar overflows, the timing system of the HXD is designed to ensure that the processing and waiting time from the detection of an event to its telemetry output is less than 32 s, which is the coverage of “EventTime”. The algorithm of the HXD time assignment is implemented in the analysis tool, “hxd-time”, in the HEASoft package.

3. Estimation of errors in the time assignment

3.1. Effects of Spacecraft Location Errors

As described in section 2.1, the first step of the time assignment of Suzaku is affected by systematic errors in a determination of orbital parameters. Due to these errors, the instantaneous spacecraft location has uncertainties of about 10 km, whereas errors along the line-of-sight from the ground station to the satellite are much smaller, about 1 km. Therefore, the air propagation delay of the telemetry can be calculated to an accuracy of 3 μ s or so.

In astrophysical analyses of the HXD data, the orbital parameters of Suzaku are also used for another purpose; to perform so-called barycentric correction, or to convert the arrival times of photons from celestial bodies for light-travel times between Suzaku and the center of gravity of the Solar system. We have developed such a tool, named *aebarycen* in HEASoft package, to perform the barycentric correction on each event. The correction has three steps; first, using the orbital parameters of the spacecraft, event arrival times are converted to those to be measured at the geodetic center of the Earth, second, to the center of the Sun, and finally to the center of the Solar system. The largest conversion in these steps comes from the second one, which is an order of a few hundred s (maximum 8 min corresponding to the traveling time from the Sun to the Earth), although it can be calculated accurately from the position of the object and the observation date (i.e., the position of Sun and the earth). Systematic errors in the overall barycentric correction mainly arise in the first step; the errors of ~ 10 km associated with the satellite position causes the barycentric times to be uncertain by ~ 30 μ s.

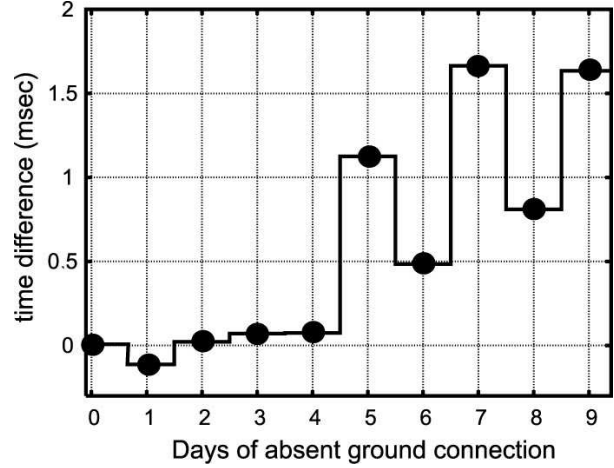


Fig. 4. Time assignment errors introduced to the Crab Nebula data taken on 2003 August 26, when the TI to UTC calibration in the off-line data processing is purposely discarded for various length of time. The horizontal axis shows duration without ground contacts. The attitude of the satellite was changed on the fourth day in the plot to observe another object.

3.2. Effects of Temperature Variation on the Clock

At the second step of the time assignment of data packets (section 2.1), namely in the conversion of TIs to UTCs, the clock frequency must be corrected for temperature-dependent drifts. Although the clock stability after the correction was confirmed to be 4×10^{-9} in a pre-flight test (section 1), the relation between the drift rate and DP temperature, used in this correction, was calibrated only on ground, and needs an in-orbit reconfirmation. Before detailed studies in section 4, figure 3 shows an example of expected clock drifts to be accumulated in ~ 100 minutes; the drift is thus predicted to be typically about 20–40 μ s, which introduces systematic errors in the absolute timing by the same order of magnitude in the worst case.

3.3. Effects of Intervals between Ground Contacts

The cross-comparison of TIs to UTC values are performed only at ground contacts. Consequently, systematic errors associated with the time-assignment on packets depend mainly on the time intervals between adjacent contacts. The operations at the down-link station are sometimes canceled due, e.g., to launch campaigns of new satellites, or typhoons attacking the local area. To emulate the effect of loss of ground contacts, we intentionally, in an off-line data analysis, omitted the TI vs. UTC cross calibrations for a given length of days. Then, as shown in figure 4, the cumulative timing errors increased as the assumed duration without ground contacts get longer.

4. Results of In-Orbit Timing Calibrations

4.1. Time Assignment of Random Signals

As described in section 2.2, the data acquisition system of the HXD is designed to avoid internal data buffering for longer than 32 s. Any failure in the design would produce

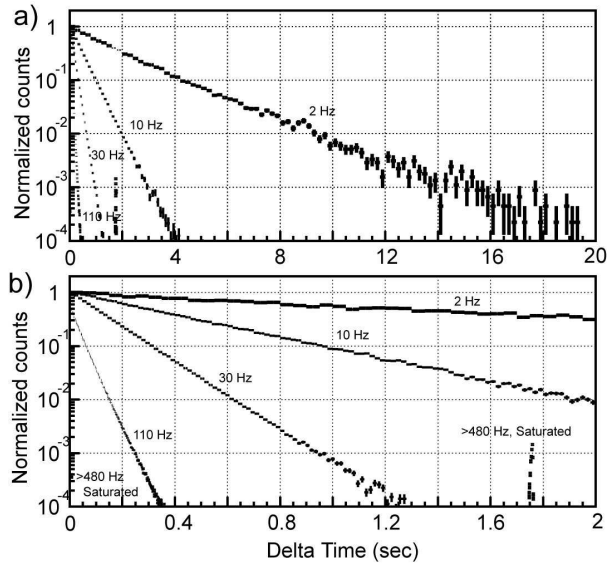


Fig. 5. Examples of time-interval histograms of HXD WEL events at various counting rates, 2, 10, 30, 110, and >480 Hz. The 2 Hz dataset was derived from cleaned PIN events acquired on 2005 September 14 for 21 ks, and those of 10 and 30 Hz are from GSO events of Crab observation on 2007 March 20 when the telemetry mode was set for bright sources. Only a quarter of GSO events were sent to telemetry when the 10 Hz data were acquired. The 110 Hz data set was taken on 2006 May 17 for an exposure of 76 ks (observation of Lockman Hole), and those with >480 Hz was taken on 2005 September 14 during a daily health-check operation of the PSD function (Takahashi et al. 1998; ?) in which the on-board background rejection was temporally disabled.

artifacts in “time-interval” spectra (occurrence distributions of time difference between adjacent events), which would normally be a simple exponential when the signals are random. We selected in-orbit unfiltered-WEL events (i.e., which include all of PIN, GSO, and BGO events) at various counting rates, and produced time-interval histograms. As shown in figure 5, the derived spectra all exhibit straight lines in semi-log plots, except when the telemetry is saturated. When the HXD event rate becomes so high that the required data flow rate exceeds the telemetry limit, some packets are discarded on HXD-DE every fixed time (1, 2, 4, or 16 s when the data rate of the telemetry is Super High, High, Medium, or Low), and thus, an artificial structure appears in the time-interval spectra; this is seen at 1.7 s in figure 5 for the 480 Hz case.

4.2. Timing Verification with Periodic Signals

In the initial performance verification phase of Suzaku, we observed four pulsars listed in table 2. After the barycentric corrections (section 3.1), we have successfully detected the periodic signals from them, as shown in periodograms of figure 6. As already reported for several of them (Terada et al. 2006; Enoto et al. 2008), pulse profiles of these objects observed with the HXD are consistent with those measured in previous observations. The pulsations were also detected in the GSO data of these

objects. As shown in figure 7, energy spectra of the pulse component of PSR1509-58 can be reproduced by a single power-law model up to 300 keV or higher. The best-fit parameters are a photon index of $1.55^{+0.10}_{-0.09}$ and X-ray flux of $4.86^{+0.34}_{-0.48} \times 10^{-10} \text{ erg s}^{-1} \text{ cm}^{-2}$ in the 10 – 300 keV band. Thus, the HXD has a capability of detecting periodic signals in the energy band of 10 to a few hundred keV, from an object of 100 mCrab intensity only in a 45 ks exposure.

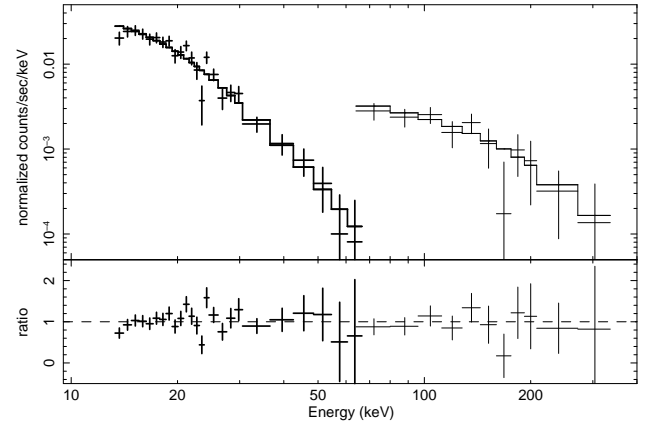


Fig. 7. Pulsed-component spectra of PSR1509-58, obtained with the HXD by subtracting the pulse-bottom spectra from the one at the pulse peak, referring to the pulse profile shown in figure 6. The pulse peak and bottom is defined as a pulse phase of 0.0 ± 0.2 and 0.5 ± 0.3 , respectively. The data of PIN and GSO are shown in crosses, and best fit power-law model of them are shown with solid lines. The fitting ranges are 13–70 keV and 70–300 keV band for the PIN and GSO data, respectively.

4.3. Relative and Absolute Timing Calibration with Crab

As listed in table 2, we observed the Crab pulsar 47 times for several calibration purposes including timing, absolute flux, energy spectral shape, and angular response. Among the 47 observations, we selected 23 observations, which have exposure longer than 20 ks, and good statistics to show high power in the period search. As plotted in figure 8 (top), the pulse periods measured on these 23 observations reveal the well known spin-down trend of the Crab pulsar, which is determined to be $4.16 \pm 0.02 \times 10^{-13} \text{ s s}^{-1}$ with our data. This rate agrees with the results by continuous radio-monitoring observations (Lyne et al. 1993). In addition, as shown in figure 8 bottom, the individual periods obtained by the PIN datasets agree with those measured by the Jodrell Bank Radio observatory (Lyne et al. 1993) within 1.0 ns. The error bars shown here, at 1σ , were determined in reference to Larsson (1996) considering higher harmonics.

The absolute timing of the HXD can be verified by comparing arrival times of the main pulse of the Crab with those indicated by the Jodrell Bank Radio ephemeris (Lyne et al. 1993). Thus, on 2007 March 20-21, we have performed a simultaneous observation of Crab with other X-ray missions, *RXTE*, *INTEGRAL*, and *Swift*. The net

Table 2. Pulsars observed with Suzaku.

Name	Period	Observation date [†]	Net Exposure of the HXD
Crab	33.58ms	2005/08–2007/03	200ks (47 observations)
PSR1509-58b	151.3533 ms	2005/08/23	44.6 ks
Hercules X-1	1.237 s [‡]	2005/10/05,2006/03/29	30.7 and 34.4 ks
A0535+262	103.375 s	2005/09/14	21.3 ks

[†] In the format of year/month/day; [‡] Without correcting for the Doppler shifts due to binary motion of the object.

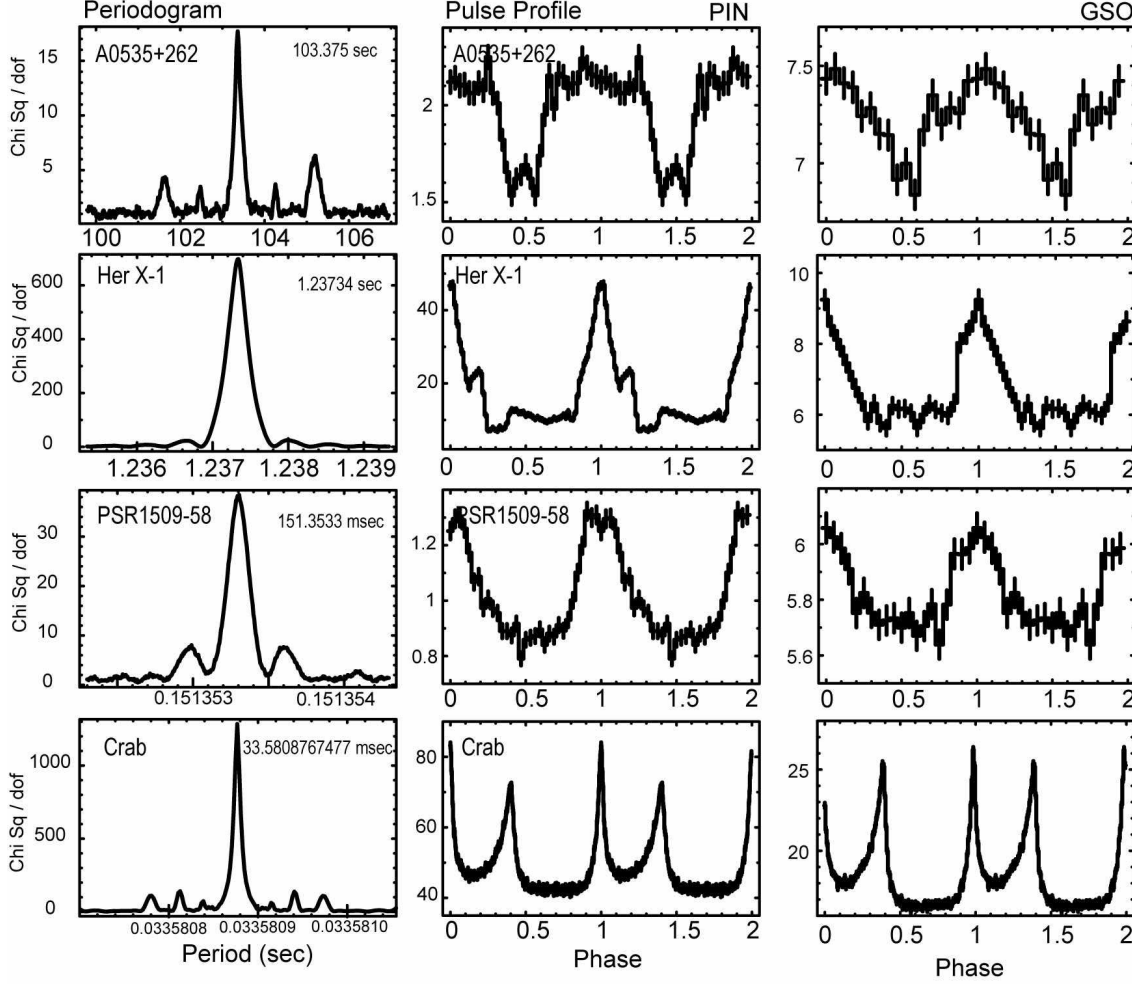


Fig. 6. Periodograms and pulse profiles of A0535+262, Hercules X-1, PSR1509–58, and the Crab pulsar, observed with the HXD. Left panels show the periodograms, obtained by PIN in the 10–70 keV band. The vertical axis means a chi-squared value when a light curve, folded at trial periods shown in the horizontal axis, is fitted by a constant. Middle and right panels show pulse profiles by PIN and GSO, respectively, folded at the periods determined by PIN in the left panel.

exposure of the HXD was 41.1 ks, of which the overlaps with *RXTE*, *INTEGRAL*, and *Swift* were 14.5, 81.3, and 24.8 ks, respectively. The pulse profiles of Crab taken in the campaign are summarized in figure 9. Thus, the X-ray pulses arrive systematically earlier by $\sim 340\text{--}500\ \mu\text{s}$ than those in the radio band. This reconfirms a previous report by Rots *et al.* (2004), who measured the same quantity as $344\ \mu\text{s}$ with *RXTE*, and discussed possible astrophysical implications. From cross-correlation studies among pulse profiles, the main pulse measured with the Suzaku HXD precedes, by 0.003 ± 0.001 , 0.002 ± 0.001 ,

0.002 ± 0.001 , 0.002 ± 0.001 phases, those from the *RXTE* PCA, the *HEXTE*, the *INTEGRAL* IBIS ISGRI, and the *Swift* BAT, respectively. Thus, in this observation, the absolute timing with the HXD systematically leads by about $160\ \mu\text{s}$ those of the other X-ray instruments.

The time difference between the HXD and the other high energy experiments in the Crab simultaneous observation is consistent with systematic errors in the HXD timing. We estimate these systematic errors by comparing several HXD-measured arrival phases of the main pulse of Crab, with the Jodrell Bank Ephemeris, as shown in fig-

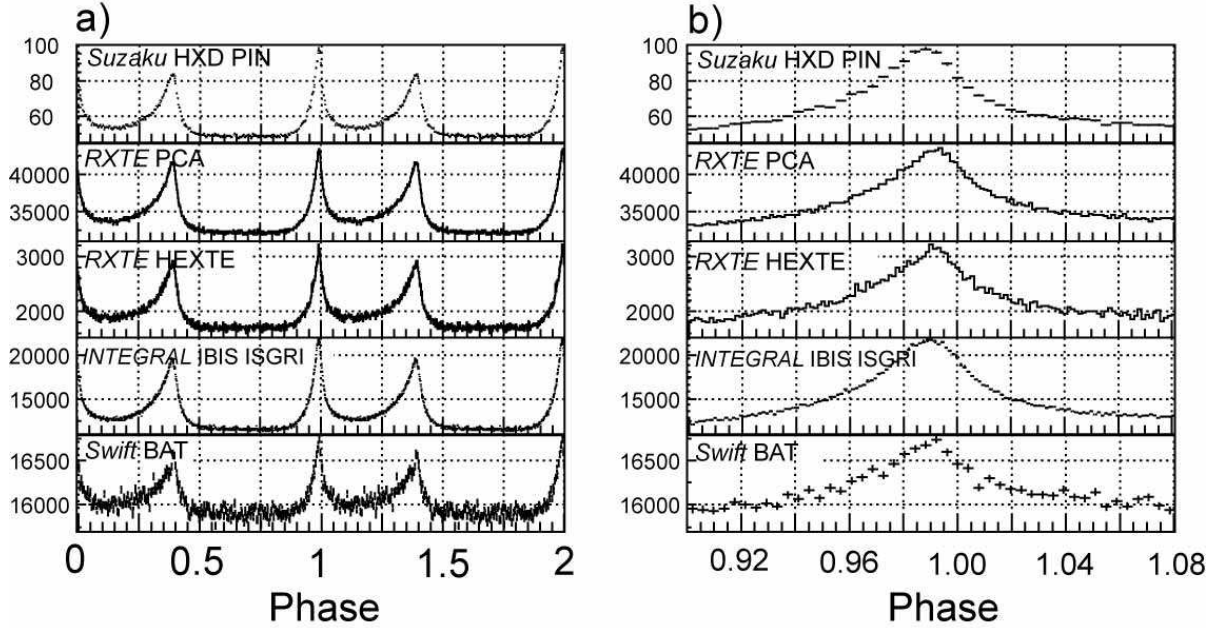


Fig. 9. X-ray pulse profiles of the Crab pulsar obtained by the Suzaku HXD PIN (10–70 keV), the RXTE PCA (2–30 keV), the HEXTE (20–100 keV), the INTEGRAL IBIS ISGRI (20–100 keV), and the Swift BAT (15–350 keV) from top to bottom. The right panel shows the same plot but expanded to reveal details over the phases of 0.91 to 1.08. The period is 0.03360091293 s, \dot{P} is $4.20506179 \times 10^{-13} \text{ s s}^{-1}$, and the phase 0 corresponds to the arrival time of the first pulse obtained by the Jodrell Bank Radio observatory; MJD 54179.4400000438310185391 (where MJD in Terrestrial Time).

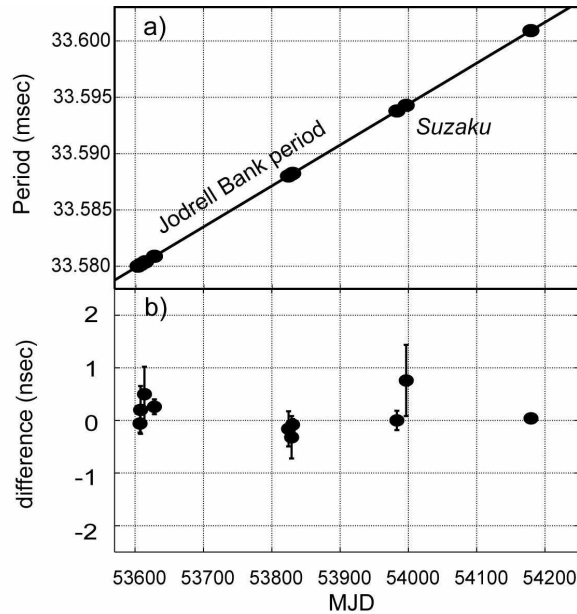


Fig. 8. Pulsation periods of the Crab pulsar obtained by the HXD, compared with the radio measurements (the solid line) at the Jodrell Bank observatory (Lyne et al. 1993). The residuals are shown in the bottom panel. Error bars were determined by the method by Larsson (1996) with a single iteration, and refer to 1 sigma.

ure 10. The averaged phase of 0.993 is consistent with the other X-ray instruments on the simultaneous observation. Since the phases measured with the HXD thus

scatters by about 0.010 ± 0.005 phases (with 90 % error), the total systematic errors in the absolute timing of the HXD is concluded to be about $360 \pm 150 \mu\text{s}$. If we discard datasets taken with the 20m antenna system or those obtained across a lack of ground contacts for > 1 day, the dispersion reduces to $270 \pm 130 \mu\text{s}$. Although the particular case of figure 9 revealed a systematic offset (by $\sim 160 \mu\text{s}$) of the HXD timing from those of other instruments, this is within the estimated systematic error of $\sim 270 \mu\text{s}$. On the other hand, the phases of the main pulses stay within 0.005 phases within an observation, implying the stability of arrival time measurements is within 1.9×10^{-9} .

5. Conclusion

Systematic errors on the HXD timing due to spacecraft location errors, the temperature-dependence of the on-board clock, and between ground contacts (if it is less than 5 days), are ~ 30 , < 40 , and $< 150 \mu\text{s}$, respectively. Using the Crab pulsar, we confirmed the stability of the relative timing to be $1.9 \times 10^{-9} \text{ s s}^{-1}$ per day. The systematic errors on the HXD absolute time have been calibrated as about $360 \mu\text{s}$ or higher ($\sim 270 \mu\text{s}$ under daily ground contacts with the 30m antenna system).

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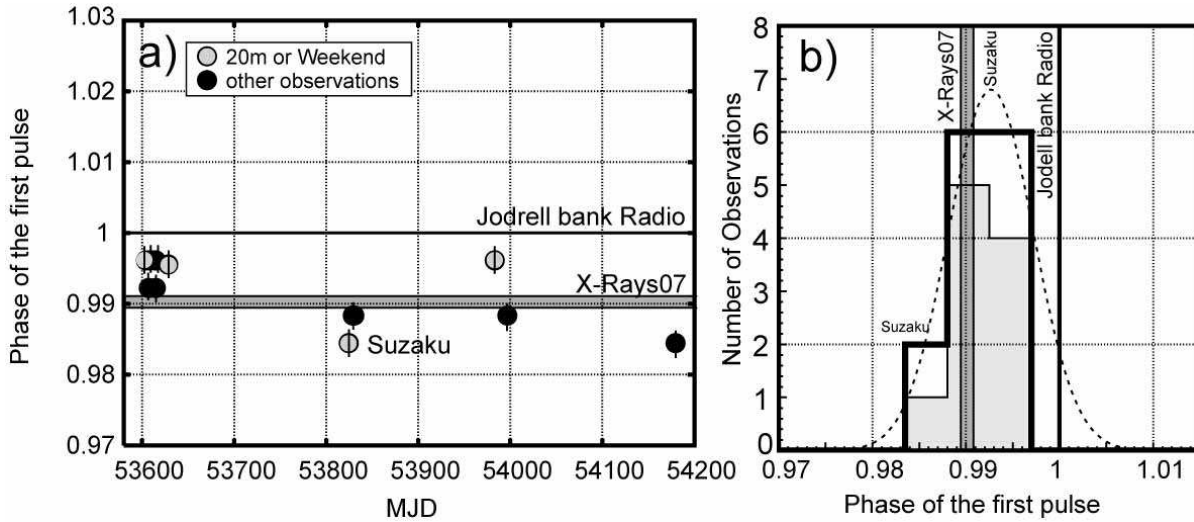


Fig. 10. (Left) Phases of the first pulse of Crab obtained by HXD PIN, compared to the Jodrell Bank ephemeris (Lyne et al. 1993). Phase 0.0 corresponds to the barycentric arrival time of the radio pulses determined by the Jodrell Bank observatory, and the Suzaku data are shown in circles. Open-gray circles show results when the observation was done with the 20m antenna or across a lack of ground contacts for > 1 day. Filled circles were taken with the 34m system on working days. The phases tagged as “X-ray 07” are determined in figure 9 by the X-ray missions in the simultaneous observation on 2007 May. (Right) A distribution (thick histograms) of the 14 phase measurements of the Crab pulse made with the HXD, compared with the radio ephemeris. The shaded histogram shows a distribution of 10 selected observations, which are taken with 34m antenna under daily ground contacts. The dashed curve is a Gaussian fitted to the distribution of all 14 phase measurements.

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